Analysis Topic: Analyst: Report created on: Document version:

#### Runway Overrun at Sao Paulo-Congonhas, 2007-07-17

*Bernd Sieker Thu Sep 06 16:15:00 CEST 2007 Document version unknown* 

# **INCIDENT ANALYSIS**

Incident description: An Airbus A320 operated by TAM Linhas Aereas overran the runway ot CGH

## Legend of Factorshapes



### Why-Because Graph



Timeline of Events

## Factor List - Overview

- 1 A/C overruns runway at approx. 100kts
- 2 A/C decelerates at 1.26m/s^2 instead of planned 3.0m/s^2
- 3 Low friction force from wheel brakes
- 4 Ground Spoilers do not deploy
- 5 Autobrake inoperational
- 6 TCU sends signal: thrust lever #2 at CL detent
- 7 Manual braking is delayed
- 8 Runway is wet
- 10 F/C did not expect ground spoilers not to deploy
- 11 F/C did not expect autobrake not to operate
- 12 F/C do not commence "LOSS OF BRAKING" procedure immediately
- 14 Runway LDA ~1880m and no RESA
- 15 Ground spoiler extension logic requires all thrust levers to be at or near idle or at reverse
- 16 A/THR systems maintains forward thrust on eng #2
- 17 A/THR system disconnects
- 18 Thrust on eng #2 freezes at approx EPR 1.18
- 19 A/THR system stays active in speed mode
- 20 Wheel brake application is delayed
- 21 Reverse thrust is selected on eng #1
- 22 A/C touches down at approx. 140 kts
- 23 F/C decide to use autobrake
- 24 Lower-than-expected aerodynamic drag
- 25 Lower-than-expected wheel brake deceleration
- 26 Persistent significant forward thrust
- 27 Significant lift from wings
- 28 Lowered friction coefficient
- 29 Low downforce on wheels
- 30 Low normal forcee
- 31 Runway surface is ungrooved
- 33 Design of A/THR control logic
- 34 Manual braking is not envisaged
- 35 Discrepancy between EPR values used by FADEC and by FMGC is larger than 0.15 for longer than 1.8s
- 36 Thrust lever angle sensors read thrust lever #2 at CL detent
- 37 EPR value becoms NCD
- 38 FADEC reduces EPR value used to ~0.75
- 39 Thrust reverser of engine #1 deploys
- 40 FMGC uses last valid EPR reading (~1.00, corresponding to IDLE thrust)
- 41 Hydroplaning
- 42 Thrust lever #2 remains at CL detent
- 43 Ground Spoilers extended annunciation does not appear

### Factor List - Details

#### A/C overruns runway at approx. 100kts 1 Type of Factor: Event Date/Time: Actors involved: Annotation: 2 A/C decelerates at 1.26m/s^2 instead of planned 3.0m/s^2 Type of Factor: Event Date/Time: Actors involved: Autobrake MED was selected, which sets a target deceleration of 3.0m/s^2 Annotation: 3 Low friction force from wheel brakes Type of Factor: Event Date/Time: Actors involved: Annotation: Ground Spoilers do not deploy 4 Type of Factor: Event Date/Time: Actors involved: Annotation: 5 Autobrake inoperational Type of Factor: Event Date/Time: Actors involved: Annotation: TCU sends signal: thrust lever #2 at CL detent 6 Type of Factor: Event Date/Time: Actors involved: Annotation: 7 Manual braking is delayed Type of Factor: Event Date/Time: Actors involved: Annotation: Runway is wet 8 Type of Factor: State Date/Time: Actors involved: This autothrust disconnect mechanism is described in the Incident report of the Taipei-Sungshan-Overrun. It is assumed to be similar in this incident, pending confirmation from AI. Annotation: 10 F/C did not expect ground spoilers not to deploy Type of Factor: Assumption Date/Time: Actors involved: Annotation: 11 F/C did not expect autobrake not to operate Type of Factor: Assumption Date/Time: Actors involved:

Annotation:

#### 12 F/C do not commence "LOSS OF BRAKING" procedure immediately Type of Factor: UnEvent Date/Time: Actors involved: "LOSS of BRAKING" is a so-called "memory item", and must be applied without delay, and Annotation: without referring to paper. Runway LDA ~1880m and no RESA 14 Type of Factor: State Date/Time: Actors involved: Annotation: Ground spoiler extension logic requires all thrust levers to be at or near idle or at reverse 15 Type of Factor: State Date/Time: Actors involved: Annotation: A/THR systems maintains forward thrust on eng #2 16 Type of Factor: Process Date/Time: Actors involved: Autothrust tries to maintain approach speed ("VAPP") Annotation: 17 A/THR system disconnects Type of Factor: Event Date/Time: Actors involved: Annotation: 18 Thrust on eng #2 freezes at approx EPR 1.18 Type of Factor: Event Date/Time: Actors involved: Annotation: 19 A/THR system stays active in speed mode Type of Factor: Event Date/Time: Actors involved: Annotation: Since not both thrust levers were set to IDLE, autothrust remains active. Wheel brake application is delayed 20 Type of Factor: Event Date/Time: Actors involved: Annotation: 21 Reverse thrust is selected on eng #1 Type of Factor: Event Date/Time: Actors involved: Annotation: 22 A/C touches down at approx. 140 kts Type of Factor: Event Date/Time: Actors involved: Annotation:

23 F/C decide to use autobrake Type of Factor: Event Date/Time: Actors involved: Annotation: 24 Lower-than-expected aerodynamic drag Type of Factor: Process Date/Time: Actors involved: Annotation: 25 Lower-than-expected wheel brake deceleration Type of Factor: Process Date/Time: Actors involved: Annotation: 26 Persistent significant forward thrust Type of Factor: Process Date/Time: Actors involved: Annotation: 27 Significant lift from wings Type of Factor: Process Date/Time: Actors involved: Annotation: Lowered friction coefficient 28 Type of Factor: State Date/Time: Actors involved: Annotation: 29 Low downforce on wheels Type of Factor: Process Date/Time: Actors involved: Annotation: 30 Low normal forcee Type of Factor: Process Date/Time: Actors involved: Annotation: 31 Runway surface is ungrooved Type of Factor: State Date/Time: Actors involved: Annotation: 33 Design of A/THR control logic Type of Factor: State Date/Time: Actors involved: Annotation: Manual braking is not envisaged 34 Type of Factor: Assumption Date/Time: Actors involved: Annotation:

35	Discrepancy between EPR values used by FADEC and by FMGC is larger than 0.15 for longer than 1.8s	
	Type of Factor:	Event
	Date/Time:	
	Actors involved:	
	Annotation:	This autothrust disconnect mechanism is described in the Incident report of the Taipei- Sungshan-Overrun. It is assumed to be similar in this incident, pending confirmation from AI.
36	Thrust lever angle sensors read thrust lever #2 at CL detent	
	Type of Factor:	Assumption
	Date/Time:	
	Actors involved:	
	Annotation:	
37	EPR value becoms NCD	
	Type of Factor:	Event
	Date/Time:	
	Actors involved:	
	Annotation:	NCD - "No Computed Data"
		This autothrust disconnect mechanism is described in the Incident report of the Taipei- Sungshan-Overrun. It is assumed to be similar in this incident, pending confirmation from AI.
38	FADEC reduces	EPR value used to ~0.75
	Type of Factor:	Event
	Date/Time:	
	Actors involved:	
	Annotation:	This autothrust disconnect mechanism is described in the Incident report of the Taipei- Sungshan-Overrun. It is assumed to be similar in this incident, pending confirmation from AI.
39	Thrust reverser of	of engine #1 deploys
	Type of Factor:	Event
	Date/Time:	
	Actors involved:	
	Annotation:	
40	FMGC uses last	valid EPR reading (~1.00, corresponding to IDLE thrust)
	Type of Factor:	Event
	Date/Time:	
	Actors involved:	
	Annotation:	This autothrust disconnect mechanism is described in the Incident report of the Taipei- Sungshan-Overrun. It is assumed to be similar in this incident, pending confirmation from AI.
41	Hydroplaning	
	Type of Factor:	Assumption
	Date/Time:	
	Actors involved:	
	Annotation:	
42	Thrust lever #2 r	remains at CL detent
	Type of Factor:	Assumption
	Date/Time:	
	Actors involved:	
	Annotation:	
43	Ground Spoilers	extended annunciation does not appear
	Type of Factor:	UnEvent
	Date/Time:	
	Actors involved:	
	Annotation:	

Actor List